# Laser Powder Bed Fusion and Direct Laser Deposition of Metals and Alloys: A Review of Developments, Process Insights, and Ni-Based Superalloy Case Studies

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#### **Abstract**

Additive manufacturing (AM) of metallic parts has gained significant attention in recent years due to its ability to produce components without traditional tooling such as molds, melting furnaces, or extensive raw material preparation. Its unique capability to fabricate complex geometries has revolutionized part design and enabled substantial weight reduction. This review first outlines the development trajectory of metalbased AM, with a particular focus on laser-based fusion methods, including Laser Powder Bed Fusion (LPBF) and Direct Laser Deposition (DLD). Understanding this evolution helps researchers identify both the capabilities and limitations of AM technologies, thereby enhancing their application in areas such as prototyping, mass production, and repair. Each metal possesses unique physical and chemical properties, which often make traditional manufacturing methods more challenging—especially for alloys with high strength, hardness, or temperature resistance. In this context, the review then focuses on nickel-based superalloys (NBSAs), which are widely used in high-temperature and high-stress environments but are particularly difficult to process using conventional techniques. Their application serves as a representative case study for evaluating the performance and feasibility of AM techniques for advanced materials. Furthermore, the future prospects of AM are discussed, including advancements in monitoring systems, integration of machine learning, and the development of AM-specific alloys. As a novel aspect, this work compares LPBF and DLD in terms of their advantages, limitations, and resulting material properties, along with a comparison to traditional manufacturing methods such as casting and wrought processing.

**Keywords:** Additive manufacturing, Nickel-based superalloys, Laser Powder Bed Fusion, Direct Laser Deposition, advanced materials

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#### I. Introduction

Additive manufacturing is one of the advanced production methods, considered one of the precursors of the fourth industrial revolution. This type of production has already replaced many traditional methods [1, 2]. All kinds of polymer, metal, and ceramic parts can be produced using AM. In the industry, several production methods could be used simultaneously to produce a piece. For instance, to produce a metal part by precision casting method, a polymer model of the part must be carved first. Then, molding is carried out using ceramic materials, and after producing molten steel, the casting process is completed, and finally, machining is performed. However, in the AM, only a computerized design file of the part and an additive manufacturing machine are needed. After receiving the file, the machine considers the design as a large number of layers stacked on top of each other and then starts to make the part [3-13].

Metals are widely used engineering materials that play a key role in industries. Traditional methods of producing metal parts, such as casting, shaping, and assembly, require equipment, tools, and high costs; however, additive manufacturing greatly simplifies the manufacturing process and, in many cases, provides higher-quality parts. On the other hand, by eliminating the complexities of production, the design of parts can be optimized. Therefore, in cases such as aviation applications where the weight of parts is important, lighter parts can be produced with higher efficiency [14].

The function of additive manufacturing machines is that parts are produced by joining raw materials together. In order to join metallic materials together, with the radiation of an energy source, the particles are partially or completely melted and added to the previous ones. After passing the energy beam, the melted part immediately solidifies. The energy beam is mainly created by the source of a laser beam or electron beam. Due to the easier accessibility and use of

lasers, this energy source has become more widespread in additive manufacturing machines. The laser beam provides a clean and reliable energy source for the fabrication of metallic parts. The additive manufacturing performed using a laser energy source is named laser-based additive manufacturing (LBAM) [15-19]. LBAM offers a key advantage over electron beam methods: it can operate in an inert gas atmosphere, making it more practical and widely applicable in industrial environments. In contrast, other sources such as arc-based systems lack sufficient controllability during processing and generate a large heat-affected zone (HAZ), which can introduce defects [20, 21]. The laser source, with its small spot size, enables precise and controlled energy input, making it suitable for both manufacturing and repair processes.

Traditionally, repair of worn components was performed using welding techniques such as Tungsten Inert Gas (TIG) welding. However, these methods often create large melt pools that compromise substrate properties and induce defects like pores and cracks [22]. The evolution of AM technologies, particularly LBAM, has introduced a more reliable and controllable alternative for defect-free fabrication and repair of parts [23].

For instance, turbine components must meet strict quality standards to operate in extreme environments, and in recent years, many of these parts have been successfully produced using LBAM. This method requires only a CAD model to fabricate complex geometries, eliminating the need for traditional manufacturing tools such as expensive casting molds or forging dies. Moreover, part designs have been customized and optimized for weight reduction in aerospace applications.

Another key benefit of LBAM is its ability to integrate component production, replacing traditional assemblies made of multiple joined parts with a single, consolidated structure. Today, LBAM is used across various industries, including automotive, defense, aerospace, and energy. In the harsh

environment of turbine operation, critical components of the combustion chamber—such as swirlers [24], inserts, and sleeves—are now manufactured using LBAM. The success of this technology has extended to the production of blades, vanes [25], and dampers in gas turbines.

Overall, LBAM offers substantial industrial advantages by reducing raw material consumption, minimizing the number of parts and machining steps, and enhancing design flexibility. Notable applications include burner tips, fuel nozzles, and components in rocket engines and small aircraft engines [26-29].

The LBAM of metallic parts has attracted much attention from industrialists recently. This approach to LBAM is due to the advantages of this method over traditional methods. Some of the advantages and features of this method are as follows.

- Due to the high temperature of the melt pool, it is possible to make parts from difficult-to-machine and refractory materials.
- The layer-by-layer manufacturing process allows any complex geometric design to be produced without limitations.
- It is possible to produce parts by changing the raw material, and the microstructure of metals can be tailored. For instance, it is possible to produce nickel-based parts with a crystallized microstructure in a specific direction, <001>.
- The properties of the produced parts are uniform throughout the whole part, and common defects, such as coring and segregation, are not observed in them.
- High energy density and small HAZ, which leads to a decrease in grain size, and therefore,
   an increase in the part's mechanical properties.
- Unlike casting and wrought methods, additive manufacturing does not require different equipment and molds for each new sample and has the highest speed and lowest cost for

custom production. Therefore, additive manufacturing is currently the best option in industries, such as aviation and biomaterials, where the variety of shapes of parts is high and the number of production is small [30-35].

In other reviews, the details of AM methods for various alloys, such as NBSAs, have been thoroughly covered [36-38]. However, understanding the development process of AM methods and how they have evolved to their current state can provide deeper insights into the advantages, disadvantages, capabilities, and limitations of each method. This review was conducted with this perspective in mind, aiming to help readers make informed decisions when selecting the appropriate AM method based on the desired final quality. For instance, using powder bed methods instead of flow-based ones might result in higher-quality parts. Conversely, understanding the limitations of the powder bed method could encourage leveraging the capabilities of flow-based methods to produce integrated parts with higher production rates.

#### II. Types of Processes

The LBAM of metallic parts can be categorized into three main types: powder-bed fusion (PBF), flow-based or directed energy deposition, and sheet lamination. Figure 1 illustrates a specific classification of LBAM systems used for metals.

The LBAM techniques employ various materials, such as powder, wire, and sheet. While powder-bed-based systems exclusively use powder, flow-based systems can utilize either powder or wire [15, 35]. Sheet lamination techniques solely rely on the use of sheets [35]. This review explains the two most common methods, powder-bed and flow-based. The sheet lamination has no usage in the industry; therefore, this method was not addressed in this review.

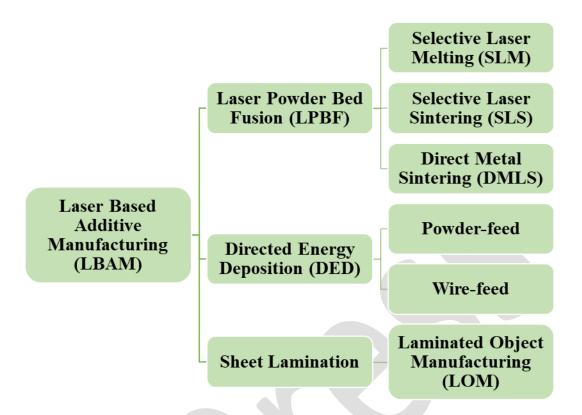


Fig. 1. Classification of LBAM techniques for metallic materials [35]

#### 1. Laser Powder Bed Fusion (LPBF)

The LPBF is a subgroup of LBAM that utilizes a high-energy laser source to selectively melt or sinter a metallic powder bed [39, 40]. The LPBF methods can be further divided into three categories, briefly described in Sections 1.1 to 1.3.

#### 1.1. Selective Laser Melting (SLM)

The SLM technique demonstrates the capacity to produce components with considerable material density, precise dimensional integrity, and desired mechanical properties. Within the SLM process, successive layers of metallic powder undergo fusion and consolidation, culminating in the development of intricate three-dimensional structures [28]. The SLM enables the fabrication of complicated components with nearly 100% density, thereby ensuring uniform characteristics across a series, obviating the need for subsequent post-processing stages [1, 11, 35].

The SLM technique provides the most typical features of powder-based additive manufacturing because of its flexibility in feedstock and shapes. Both inert argon and nitrogen gas can be utilized in the SLM process. The schematically working system of the SLM machine is demonstrated in Figure 2. Nowadays, SLM terminology is mainly known as its group head, LPBF; therefore, in the case of this review, it will be regarded. Refer to ASTM 52900-2022 for more details about additive manufacturing terminologies.

Despite the numerous advantages of LPBF, it still exhibits certain limitations in its processing compared to traditional manufacturing techniques. Due to the localized concentration of energy input, a temperature gradient mechanism is induced, resulting in plastification and subsequent generation of residual stress, ultimately leading to deformation. These residual stresses play a pivotal role in affecting dimensional accuracy and propagation of cracks, potentially causing detachment of parts from the base plate [41].

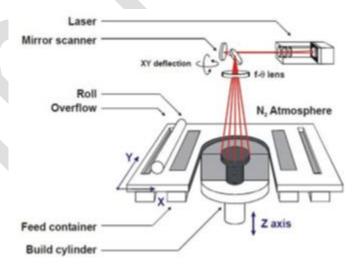


Fig. 2. Schematic representation of an SLM machine [1]

It would be suggested that the material be preheated to minimize the residual stress. The normal thickness of the layers in the LPBF process is between 20 to 100 μm. In the process, one of the Nd: YAG or CO<sub>2</sub> lasers is used as the heat source. A wide range of metallic materials can be processed by the LPBF. Some of these materials are as follows: stainless steel, aluminum, copper, iron, cobalt-chrome, titanium, NBSA, and a mixture of mentioned ones [35, 42].

#### 1.2. Selective Laser Sintering (SLS)

In the SLS process, a mix of two different powders is used. One powder is structural, and the other is sacrificial as a binder. The SLS machine works similarly to the LPBF, and the powder is different. The structural powder is a metallic material, and the sacrificial powder is a polymer. While the laser beam irradiates the powder mixture, only the polymer melts, and the structural powder remains unchanged. The fused polymer binds the structural material together and makes an integrated part. Heat treatment should be applied to remove the binder and sinter the structural powder. Thereby, the sintered part, known as the green part, is held in order of 900°C temperature. The green part has approximately 50% porosity [35]; as a result, the sintered component undergoes a transformation process through infiltration with a low melting point metal or alloy, such as copper, brass, and bronze, thereby yielding a dense composite alloy component [43]. It is noteworthy that the accuracy of the SLS process is challenging to predict as it is a function of various parameters, some of which can be mutually dependent. The parameters that have the most influence on SLS/Rapid Prototyping accuracy can be divided into three groups: pre-processing, processing, and post-processing errors [44]. Mostly, the layer thickness is between 100 to 300 µm, and SLS resolution is also in the order of 100 µm. One of the advantages of SLS is the lower consumption of energy. Therefore, the embedded laser can be each fiber, CO<sub>2</sub>, or disc laser without any specific limitation. Furthermore, the other advantage is the processing possibility of a wide range of materials, such as sand, stainless steel, and different plastics [35]. Metal powders find their exceptional applicability in the realm of SLS, as it is challenging to directly fabricate metallic components using alternative rapid prototyping, rapid tooling, or rapid manufacturing (RP/RT/RM) methods [45].

The SLS can be further enhanced through a variation known as SLS/Hot Isostatic Pressing (SLS/HIP), which introduces several advantageous features into the manufacturing process. The SLS/HIP represents a net-shape manufacturing approach that merges the inherent freeform shaping capabilities of SLS with the complete densification potential of HIP [43]. These two features are next to each other, leading to a reduction in manufacturing costs. In the mere SLS, the part has a densification of approximately 80%. After HIP treatment, a fully dense specimen can be produced. The HIP process uses an inert gas like argon, and the put sample experiences a high temperature at a high level of isostatic pressure. The details of the method, such as layer thickness and the resolution, are the same as mentioned about SLS [35]. According to Liu et al. [43], cold isostatic pressing exerts a nearly equivalent influence on the final properties of the component. This method is particularly suited for processing durable, high-strength materials like IN625 and Ti6Al4V [35].

#### 1.3. Direct Metal Laser Sintering (DMLS)/Direct SLS

The technique known as DMLS or Direct SLS involves the utilization of two distinct types of metal powders. One of these powders possesses a high melting point, serving as the structural metal, while the other features a lower melting point, fulfilling the role of a binder [33]. It is noteworthy that the DMLS process can alternatively employ a single powder with varying grain sizes, where the powder with a smaller size is fused, and coarse/structural powder will sit in it [35].

In its fundamental concept, the DMLS process closely resembles SLS. However, DMLS employs uncoated pre-alloyed metal powders as raw material, whereas SLS relies on polymers or coated metal powders [4].

Both the SLS and DMLS are suitable for tooling; DMLS does not need the time-consuming step of removing excessive binder material. This advantage makes the production process faster and more economical. Therefore, the DMLS method can be applied to produce prototype models, molds, and dies [44, 46].

#### 1.4. Comparison Between LPBF and SLS

In the LPBF process, the powder completely melts during the laser beam irradiation. This full melting manufactures a final part with high density, and its quality is comparable to that of the conventional method. The SLS parts have a high porosity volume; thereby, their final quality is decreased [35, 44, 47].

Melt infiltration and removal of the binder are two time-consuming steps. The LPBF parts do not need any specific post-processing. Therefore, if the surface roughness is ignored, the final part after the cutting from the substrate can be used in industrial applications [35, 44, 47].

Therefore, in the past few years, LBAM of metallic materials has been considered limited to LPBF and directed energy deposition (DED) processes and no longer pointed to SLS [2, 48, 49]. Moreover, in the continuation of this review, only LPBF properties are reported, and SLS reports are ignored as much as possible.

#### 2. Flow-based or Directed Energy Deposition (DED)

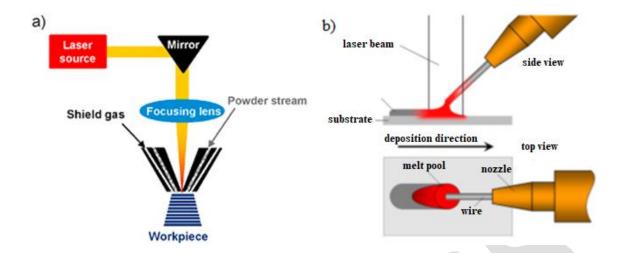
The flow-based deposition is firmed on the injection of the powder or a wire as the feeder of the process to create a metallic part. The injected wire or powder melts by a heat source. Moreover, a

laser or an electron beam can be used as the heat source. The process is called "directed energy deposition (DED)" or "direct metal deposition." If the heat source is a laser, the process is also known as "directed laser deposition" [2, 39, 50]. Due to the numerous brands that produced the DED machine, the process has taken many names. Some of the most prevalent ones are laser-engineered net shaping (LENS) [50], direct light fabrication (DLF) [51], laser consolidation (LC) [52], laser rapid forming (LRF) [53], and laser cladding (often for coatings) [35].

#### 2.1. Direct Laser Deposition (DLD) Equipment and Process

The DLD mechanism consists of the blown powder or injected wire that is introduced to a substrate and melts by a focused laser while reaching (Figure 3). The laser type is a high-power one and can be gas-CO<sub>2</sub> or fiber-Nd: YAG. In some handy-made DLDs, a CNC with a powered laser and a powder blower pump can form the machine [51-53].

Blowing of the powder or injecting the wire is carried out by a nozzle. In order to prevent the oxidation of the created melt atop the substrate, an inert gas is blown into the melt pool. Moreover, in the powder DLD process, the volume of blown powder is often several times more than what melts, i.e., a large volume of the powder blown from the nozzle remains unscraped and unused and just plays the role of the oxidation barrier. Furthermore, the inert gas for the powder-base DLD and the oxidation barrier play the role of the powder carrier [53, 54].



**Fig. 3.** Schematic drawing of the DLD process, a. powder-feed laser deposition, and b. Wire-feed laser deposition [51, 52]

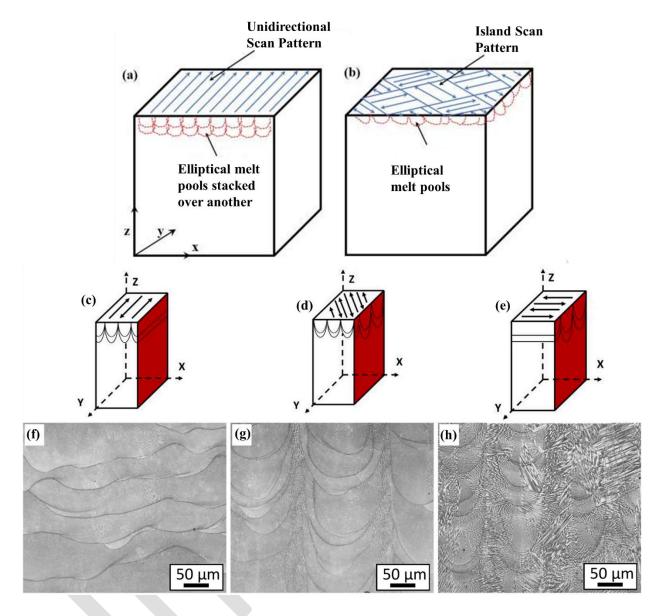
#### 3. Influence of Process Parameters and Scan Strategy on Microstructure and Properties

Process parameters significantly influence the microstructure and mechanical properties of the part. These parameters can be broadly categorized into two main groups: those affecting (1) energy density and (2) layer thickness [5, 55-58]. Any critical factor impacting part quality may be considered a process parameter, including laser source characteristics, part design, machine components (e.g., nozzle configuration), and raw material properties.

In the DLD method, the most critical process parameters are generally grouped into four main categories: laser power, laser scanning velocity, powder mass flow rate [59, 60], and hatch spacing [35, 61]. With the exception of powder mass flow rate, these parameters are also relevant in the LPBF process. However, in LPBF, the layer thickness is a key parameter that replaces the powder mass flow rate used in DLD. It is important to note that LBAM processes may include additional process parameters—for instance, laser beam focus diameter and laser standoff distance, both of which can influence build quality in DLD and LPBF methods [62].

Moreover, scan strategy or pattern is another influential factor that affects the properties of the final part (Figure 4). Various scan strategies—such as unidirectional, bidirectional, continuous, and raster/island—result in distinct material properties [63]. In certain cases, interlayer rotation is essential for achieving a dense, pore-free structure. For example, in one study (Table 1), a 30° interlayer rotation increased the density of Inconel 718 parts from 8.11 to 8.20 g/cm³ [62]. Parts produced with interlayer rotation typically exhibit a basket-weave structure, as illustrated in Figure 9 [35]. Further discussion on scan strategies is provided in Section III.4.

Also, powder characteristics are closely linked to process parameter optimization. Selecting suitable parameters requires consideration of material properties. Attributes such as powder shape, size, and distribution directly influence process behavior, including flowability (especially in DLD), laser absorption, and surface morphology [47, 64, 65].



**Fig. 4.** Effect of scanning pattern on melt pool and microstructure: unidirectional (a) and island (b) scan patterns [66]; melt pool indications influenced by scanning orientation, interlayer rotation of 67 degrees (c to h) [67]

**Table 1.** Effect of interlayer rotation on increasing the density of the part/sample [62]

| Interlayer rotation degree | Porosity (%) | Density (g/cm <sup>3</sup> ) |
|----------------------------|--------------|------------------------------|
| 0                          | 0.21         | 8.11                         |
| 30                         | 0.10         | 8.20                         |

#### 4. Comparison Between LPBF and DED

Each LPBF and DLD is used when considering the final required properties. The production speed of DLD is often higher than that of LPBF, and the surface quality of LPBF is better than that of DLD. Altogether, DLD may be offered for large parts in which the surface quality is not a limitation, and the production speed is important.

#### 4.1. Advantages of DED / DLD Over LPBF

One of the most important advantages of DLD is its high productivity speed. The deposition rate is approximately 100 gr/h. The layer thickness reaches 250  $\mu$ m, but the value for LPBF is conventionally less than 40  $\mu$ m [35].

The other advantage is its freedom in the part size, which means that the DLD, as opposed to LPBF, is not constrained to the build chamber and powder container [35]. The other unique feature of DLD is the capability to produce functionally graded materials and a single material. Moreover, DLD can be used in repairing parts, often known as laser cladding repair [49, 54, 68-70].

#### 4.2. Disadvantages of DLD Compared to Powder-Bed

The main disadvantage of the DLD process is the degraded surface quality. In DLD, the roughness is high and at least ten times higher than that of LPBF. Therefore, the final part must frequently be machined and surface treated [2, 35, 48].

Some defect formations are prevalent in DLD, which are as the following:

- Cracking and distortion initiated from a high cooling rate
- The porosity derived from the powder contamination and the gas entrapment
- Lack of fusion (LOF) and weak bonding between the layers
- Hatch line defects owing to improper process planning

#### 4.3. Powder-Bed Advantages Over Flow-Based Methods

for parts having overhangs, LPBF is better than DLD. The other advantage of LPBF is its low residual stress. Altogether, thermal tension in the additive manufacturing processes is high because of fast repeated melting and cooling cycles. Nevertheless, this residual stress in LPBF is less than DLD because the powder bed inhibits the fast cooling of the melt section and exit of the heat [2]. Better surface quality and higher dimensional accuracy are the other basic advantages of LPBF over DLD. The purpose of applying additive manufacturing is to produce delicate parts with complex geometries. Therefore, owing to LPBF's superiority, it will be advised when high accuracy is required [71, 72].

In powder bed processes like LPBF, the powder bed prevents the melt from falling. Accordingly,

#### 4.4. Limitations of LPBF/Powder Bed Compared to DLD

Powder bed methods are constrained to the build chamber. However, in DLD, the process can be designed so that the part size can vary freely. The deposition rate of LPBF is lower than that of

DLD; nevertheless, it is noteworthy that in new LPBF machines, the total deposition rate has increased with the increase in the number of lasers. For instance, there are simultaneously 12 active lasers in some new LPBF machines [35, 47, 48, 73]. The LPBF process is more sensitive to the size of the powder. If the powder particles are large, the accuracy is challenged. If the powder is fine, the tendency to agglomeration increases and the flowability faces difficulty [47].

#### 5. Monitoring

In past years, monitoring has been seriously discussed to find the onset of the defects in the serial production of the parts. In additive manufacturing processes, the stability and repeatability of the procedure are challenges. Some devices and software were designed to detect anomalies during the process quickly. The conventional tools to help the monitoring and detection of the distorted part are as follows [35, 64, 72, 74-79].

- Infrared (IR) imaging
- Ultraviolet (UV) imaging
- X-ray imaging
- Charge-coupled device (CCD) video imaging
- Photodiode
- Pyrometer
- Ultrasonic wave generator
- Complementary metal oxide semiconductor (CMOS) camera

#### 5.1. Powder Bed Fusion Process Monitoring

In the PBF, the melt pool, the layer of the powder bed, the manufactured slice, and the underscanning tracks can be evaluated. Another important issue in powder bed monitoring is the spattering of the powder or melt. Since 2011, galvanic scanners have helped monitor the powder bed. The detection is based on measuring two angles and the wavelength differences resulting from the instant temperature. The detection normally faces displacement problems, which can be solved using two-dimensional sensors [75, 79-85]. Currently, most new monitoring systems are based on non-contact systems consisting of optical, thermal, and acoustic detectors. The mentioned devices and the rained computers powerfully detect the anomalies [64, 77].

#### 5.2. Monitoring of Flow-Based Processes

The DLD process monitoring can be carried out in every step of powder delivery, melt pool, and layering.

#### Monitoring of Powder Delivery Rate

A photodiode is set to measure the volume of the powder exiting from the nozzle. The more powder particles pass through the nozzle, the less light reaches the photoelectric sensor. Furthermore, the optical and acoustic sensors are promising measurement devices in the delivery rate step [30, 35, 78, 86, 87].

#### Monitoring of Melt Pool and Layer Morphology

Melt pool monitoring is often based on thermal manners. A pyrometer and an IR sensor coupled with a CMOS or CCD camera can evaluate the melt pool morphology. Proper monitoring of the melt pool can evaluate and improve the geometrical integrity, as well as the microstructural and mechanical properties of the under-manufacturing part. Since the melt pool shrinks, lengthens, and splashes, its morphology during the laser scan is completely unstable. This instability makes it difficult to evaluate and monitor the process. In order to monitor the unsteady melt pool, thermophysical equations like Rothensal's analytical solution are mainly applied. Overall, melt

pool characteristics, such as peak temperature, length along a certain axis, and total area, are some of the process signatures [2, 35, 88, 89].

X-ray tomography is proposed as a new and reliable method to monitor layer morphology. The X-ray setup characterizes, quantifies, and identifies the layer morphologies and anomalies [90].

#### 6. Machine Learning

Computers can help to detect anomalies in each step of the manufacturing process. This type of computer assistance is known as machine learning (ML). In some cases, ML is coupled with monitoring tools to detect and obviate process defects. There are many toolsets accessible for employment in the realm of image processing within the domain of ML, encompassing both freely available and commercially licensed alternatives. Noteworthy examples include the MATLAB Computer Vision Toolbox and the C++/Python OpenCV libraries. Overall, ML can improve all steps of the manufacturing process, from parameter settings to quality control. Numerous sciences like metallurgy, electronics, physics, and mechanics are engaged in additive manufacturing technology. Moreover, in each field of the process, many parameters and variables may change [91-95].

In order to optimize an additive manufacturing process, the computer takes the variation range of the parameters and assumes a step for each parameter/item, then compares the results and suggests the best one. For example, in one additive manufacturing process, the parameters and steps are according to Table 2.

Therefore, there are many degrees of freedom in normal additive manufacturing work. The computer considers all these degrees and offers the optimized one in supervised or unsupervised situations [96-100]

Table 2. Case study illustrating a method to optimize the process using design of experiments

| Parameter/Item            | Range/states                  | Step size  | Number of states = (range size /step size) + 1 |  |
|---------------------------|-------------------------------|------------|--|--|
| Power                     | 100-300 w                     | 20 w       | 11   |  |
| Scan speed                | 500-1500 mm/s                 | 100 mm/s   | 11   |  |
| Hatch Distance            | 30-80 µm                      | 10 μm      | 6  |  |
| Layer Thickness           | 20-60 μm                      | 10 µm      | 5  |  |
| Next Layer Rotation       | 0-90 degrees                  | 15 degrees | 7  |  |
| Amount of Recycled Powder | 0-100 percent                 | 20 percent | 5  |  |
| Build direction           | 0-90 degrees                  | 15 degrees | 7  |  |
| Powder Size               | 15-53 μm/ 60-80 μm            | -          | 2  |  |
| Scan strategy             | Island/Regular Back-and-Forth | -          | 2  |  |

#### III. Mechanical properties

The mechanical properties of NBSAs have been investigated for several decades. Conventionally, the superalloy specimens were produced by wrought or casting. However, after the emergence of AM, it turned into a desired method. The additive manufacturing can produce refractory materials with high-temperature applications and manufacture parts in complex shapes.

Gas-turbine blades have both specifics, complicated designs, and high-temperature materials. The blades sometimes have air paths that make their design more complicated and harder to produce. Therefore, additive manufacturing seems to be a suitable method of manufacturing these parts, which has been investigated seriously in past years [101-105].

#### 1. Various Items Influencing the Final AM-Developed Part's Quality

An extensive range of different parameters influences the properties of NBSAs. On the other hand, a wide variety of elements are used in the composition; this difference in the alloying makes the final properties more complicated. Therefore, AM-developed nickel-based parts need to optimize the process parameters, based on the added alloying elements. In the past years, much research has been dedicated to the metal additive manufacturing field, but it is not sufficient yet due to the variety of parameters, specifically for nickel.

- a. Process parameters of additive manufacturing significantly affect the specimen properties. The important parameters are scan velocity, laser power, hatch distance, and layer thicknesses [39, 106-108].
- b. The grain structure in NBSAs has an important impact on the part application. For high-temperature usage, mostly single crystal and columnar grains are used. In AM-developed parts, the structure prefers to form columnar dendrites. When the process parameters are

fixed, an equiaxed, columnar, or single-crystal structure can be determined. Some process conditions, such as build direction, preheat temperature of powder or substrate, and scan strategy, can help better tailor the desired dendrite structure [36, 37, 109-114].

#### 2. Different Types of Powders in the Additive Manufacturing Processes

Since additive manufacturing is somewhat comparable to the welding process, most research is performed around alloys with acceptable weldability, such as IN718, Waspaloy, and Nimonic263 [115]. However, it is necessary to mention that the number of usable prevalent powders is not numerous and does not have a wide variety. Therefore, published papers regarding the additive manufacturing field of the NBSAs are mostly limited to a few specific and conventional grades [35].

Mainly three types of powders are used in the additive manufacturing processes of NBSAs:

- Powders with good weldability, like IN625, which avoid cracking because of solid solution strengthening and low content of precipitates [35, 116-120].
- Non-weldable powders that are susceptible to strain-age cracking because of gamma-prime ( $\gamma'$ ) precipitates.

In these alloys, the sum of the aluminum and titanium element content is typically more than 6.4%. The IN738 and CMSX-4 are among the examples of these alloys [53, 109, 121].

• Alloys with medium weldability are also part of NBSAs. In these cases, although the gamma-double prime secondary phase ( $\gamma$ ") precipitates in the matrix, this phase opposite of  $\gamma$ ' does not face the specimen to considerably high strain age cracking. This precipitate-strengthening mechanism is used in the IN718 alloys. This balance between strength property and desired weldability has resulted in the general usage of IN718 in the additive manufacturing processes [1, 51, 122-131].

It is noteworthy that IN718 has proper weldability as well as structural and mechanical stability up to  $650^{\circ}$ C; therefore, this alloy is one of the most applicable additive manufacturing superalloys used in elevated-temperature applications, such as gas turbines and aviation engines. The IN718 is a superalloy made of nickel, iron, and chromium as the major elements in which the main strengthening precipitate is a semi-stable phase,  $\gamma$ ", with a tetragonal body-centered cubic crystal structure [115]. The microstructural properties of the manufactured specimen and precipitates play a key role in the strengthening properties. Therefore, the characterization of the type, size, and dispersion of the precipitates has significant importance. The precipitate-strengthening properties in the IN718 are determined by the heat treatment cycle. Therefore, to reach the maximum mechanical properties, the heat treatment issues are bolded [115]. In this regard, different types of precipitates of the IN718 are listed in Table 3 [132].

Table 3. Existent phases in NBSAs [132]

|                       | Phase       | Crystal system | Primary Composition                   |
|-----------------------|-------------|----------------|---------------------------------------|
| Solid Solution Matrix | γ           | Cubic          | Ni, Cr, Fe-Based                      |
|                       | γ'          | Cubic          | Ni <sub>3</sub> (Ti, Al, Nb)          |
| Intermetallic         | γ"          | Tetragonal     | Ni <sub>3</sub> (Nb, Ti)              |
|                       | δ           | Orthorhombic   | Ni <sub>3</sub> (Nb, Ti)              |
| Topological           | η           | Hexagonal      | Ni <sub>3</sub> (Ti, Al)              |
| (Intermetallic)       | Laves       | Hexagonal      | $(Ni, Cr, Fe)_2(Nb, Ti)$              |
| Carbide               | MC          | Cubic          | (Nb, Ti)(C, N)                        |
|                       | $M_{23}C_6$ | Cubic          | (Cr, Fe) <sub>23</sub> C <sub>6</sub> |

## 3. Comparison Between Additive Manufacturing and Conventional Methods in Manufacturing Steps and Resulting Microstructure

In previous sections, it was mentioned that one of the strengthening mechanisms of NBSAs is issued with secondary phase precipitates ( $\gamma'$  and  $\gamma''$ ). In order to form and optimize this strengthening, the as-manufactured specimens must be heat-treated. Notably, because of their inherent variation from conventional methods, the AM-developed parts include secondary phases with smaller sizes. These differences between precipitate sizes are derived from a small melt pool, little time for secondary phases to join each other, and thereby less coring in the additive manufacturing processes [35, 51, 53, 62, 63, 106, 121, 123].

Regarding the past sections, it is clarified that all of the manufacturing steps in nickel-based additive manufacturing processes can influence the printed part properties. These steps include the following:

- Pre-manufacturing step and use of powder and chamber preheat [121]
- Manufacturing step and the process parameters [57, 109, 133]
- Scan strategy/pattern [62, 63]
- The post-process level and usage of heat treating [134]

Currently, accurate and obvious evaluation of the NBSAs properties in layer manufacturing/additive manufacturing is complex. Microstructural variations are derived from the process method, process parameters, and part geometry. All mentioned items simultaneously play a role in the mechanical properties, but the impact of each one and its share are not fully clear. After AM/forming/casting, NBSAs is mostly put under standard heat treatment. Nowadays, almost all studies regarding post-manufacturing heat treatment are based on the same information

received from the casting part. Therefore, due to the intrinsic disparities between the additive manufacturing structure and its counterparts, previously established standards may prove suboptimal in this context [25]. [35].

Furthermore, the mechanical properties, including the structure, display a non-equality/non-uniformity in the properties owing to different build directions. Currently, there are no standard samples for the mechanical tests of AM-developed parts. The standard sample should consider important properties of the layer manufacturing, such as surface roughness, post-processing, and build direction. Considering the impact of the build direction on the microstructure, in some reports, tests are carried out by sampling in both vertical and horizontal directions [35, 135].

### 3.1. Comparison Between Properties of Additive Manufacturing and Conventional Methods

In initial comparisons between additive manufacturing (DLD/LPBF) and conventional methods (wrought/casting), it was reported that heat-treated AM parts possess at least 80 % of the tensile strength of their wrought counterparts. Although LPBF outperforms DLD in as-deposited condition, DLD exhibits the lowest room-temperature tensile strength without heat treatment.

In most case studies, AM-produced samples demonstrate significantly greater elongation, exhibiting at least 25 % more elongation than wrought specimens [35, 106, 136]. The final part properties have been further enhanced by process-parameter optimization and machine upgrades—such as powder preheating. Recent studies have shown that, with appropriate heat treatment, creep properties of AM parts can exceed those of both wrought and cast components (Figure 5) [137].

Regardless of the influence of process parameters on pore formation, each AM process has inherent capability to achieve full density. Among conventional AM techniques, LPBF currently

delivers the highest part densities (Table 4). This superior densification is a key advantage of laser-based systems over electron-beam systems and underpins their enhanced mechanical performance [138]. Consequently, LBAM parts are regarded as more reliable for industrial applications, especially under extreme conditions such as those in gas turbine components.

Table 4. Densification capability of AM methods

| cation capability | of AM methods        |                |           |
|-------------------|----------------------|----------------|-----------|
| Alloy/Element     | Relative Density (%) | Process        | Reference |
| AlSi10Mg          | 99.77±0.08           | LPBF           |           |
| Mg                | 95.28 to<br>96.13    | LPBF           | [139]     |
| Ta                | 99.6                 | LPBF           |           |
| Ti-6Al-4V         | < 99                 | LPBF           |           |
| ABD-900 AM        | 99.9                 | LPBF           |           |
| ABD-900 AM        | 98.4                 | EBPBF          | [26]      |
| ABD-900 AM        | 99.92                | EBPBF +<br>HIP |           |
| IN738LC           | 99.87 to<br>99.96    | LPBF           | [140]     |

### 3.2. Impact of Build Direction on the Mechanical Properties and its Comparison with the Casting Characteristics

The build direction directly influences the microstructure and properties of the part. In Figure 6, the two z and xy build directions are indicated [115, 141]. In the z-build direction, the specimen morphology is formed columnar parallel to the part axis. Nevertheless, in horizontal manufacturing of parts, the morphology similar to casting parts is manufactured in equiaxed form (xy-specimen in Figure 6).

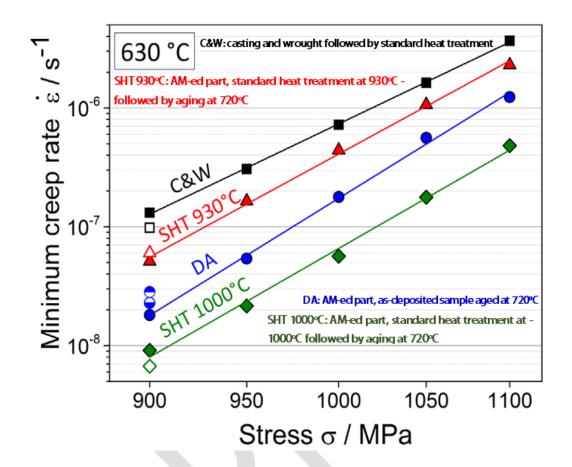
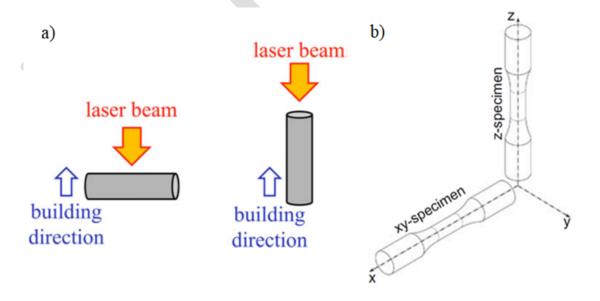


Fig. 5. Creep rate between 900 to 1100 MPa until 1% plastic deformation [137]



**Fig. 6.** (a) two various build directions in the LPBF build chamber [115], (b) schematic of the build direction concept [141]

Among the properties of NBSAs, creep is significant owing to the high-temperature applications of these alloys. Rickenbacher [141] et al. demonstrated that IN738 HIP samples manufactured in the vertical direction (z) have more creep strength compared to those manufactured in horizontal directions (xy). The creep strength of z specimens equals the minimum of the casting ones. Although the creep properties of the z specimen are better than xy, the other mechanical properties are less than xy. The comparison of tensile strength of the z specimen versus xy is according to Tables 5 and 6. This greater strength can be related to the impact of the hall-petch relation and grain boundary resistance against the dislocation movements in temperatures less than 0.5 Tm.

**Table 5.** Tensile properties of the cast and LPBF samples at room temperature (IN738) at 23°C [141]

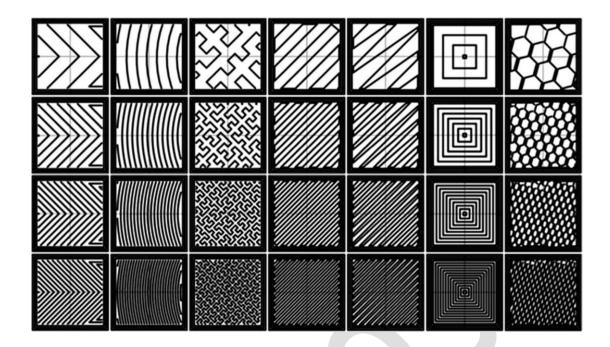
| sample            | Young's<br>modulus<br>(GPa) | σ <sub>(0.2)</sub> (MPa) | συτς (MPa) | Elongation (%) |
|-------------------|-----------------------------|--------------------------|------------|----------------|
| Casting reference | 200                         | 765                      | 945        | 7.5            |
| SLM-xy            | 233±9                       | 933±8                    | 1184±112   | 8.4±4.6        |
| SLM-z             | 158±3                       | 786±4                    | 1162±35    | 11.2±1.9       |

**Table 6.** Tensile properties of the cast and LPBF samples at 850°C (IN738) [141]

| sample            | Young's<br>modulus<br>(GPa) | σ <sub>(0.2)</sub> (MPa) | σ <sub>UTS</sub> (MPa) | Elongation (%) |
|-------------------|-----------------------------|--------------------------|------------------------|----------------|
| Casting reference | 144                         | 530                      | 710                    | 10.0           |
| SLM-xy            | 157±4                       | 610±1                    | 716±1                  | 8.0±1.2        |
| SLM-z             | 110±2                       | 503±2                    | 688±7                  | 14.2±3.9       |

#### 4. Scan Strategy Effect on the Mechanical Properties

The additive manufacturing scan strategy is an important item that highly impacts structural and mechanical properties. Many different scan patterns can be applied in part manufacturing. Scan patterns can include conventional linear and/or island patterns. In general, various forms of laser passes can be used on the specimen (Figure 7). Each of these forms induces a distinct transformation within the ultimate microstructure of the specimen. This variance in the resulting microstructure can be attributed to the fluctuation in the energy input applied to the powder and the final manufactured part. Overall, there are five common scan patterns, which can be seen in Figure 8 [35].



**Fig. 7.** Scan strategies influencing energy density and heat input, subsequently affecting microstructure and residual stress [35]

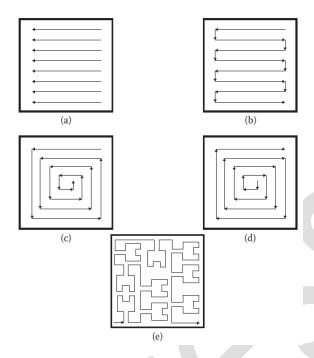
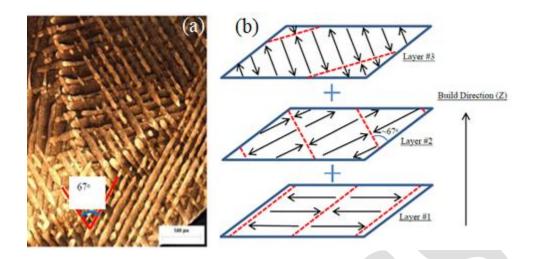


Fig. 8. Five conventional scan patterns [35]

According to scientific developments in the field of scan patterns, laser pass rotation in each layer is effective in improving the properties. After the deposition of each layer, the scan pattern is applied at a different angle to manufacture the next layer. Therefore, porosities in the part are reduced, and the sample density is near the theoretical density. The optimized angle in the literature is reported to be approximately 67°, between (30° and 90°). This strategy is also called the basket pattern [62, 63, 142, 143]; Figure 9 demonstrates the schematic image and the microstructure of a part with a basket pattern (more details are available in Section II.3).



**Fig. 9.** a) Microstructure of the part with the interlayer rotation, b) A sample schematic of the rotation strategy [63]

## 4.1. Decrement of Residual Stress Using the Island Pattern

One of the significant problems in the additive manufacturing of parts is the residual stress in the sample and the deformation of the part. In order to avoid this problem, the island scan strategy is employed. Using conventional patterns, the energy diffuses focally to the sample, resulting in residual stress and distortion. To decrease the remaining stress, the energy is diffused dispersedly to the under-manufacturing specimen (Figure 10). The island strategy is used to satisfy this need. The manner may include various states. The totality of the island pattern is that the under-process layer is considered a raster, similar to a chess board. Each cell grid of the board is selected randomly and scanned by the laser beam to be manufactured. The random application of the energy to each cell causes the avoidance of focused energy inserted into one area of the part. The dispersed application of the energy inhibits part distortion and residual stress [144-148].

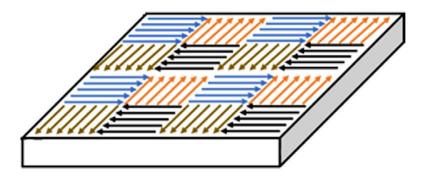


Fig. 10. Schematic of island scan strategy [148]

### 4.2. Hatch Spacing Distance

Hatch spacing is the distance between adjacent laser scan tracks. In general, hatch distances are chosen to ensure approximately 30% overlap between passes. For instance, if the melt pool diameter is 80 µm, a hatch spacing of around 60 µm may be used to achieve this overlap [62, 149]. As presented in Equation 1, increasing the hatch spacing directly reduces the input energy. Improper selection of hatch spacing can lead to bonding defects, LOF, altered cooling rates, residual stresses, interlayer porosity [150, 151], surface roughness [36], and cracking [109]. In fact, most defects related to energy density can be influenced by hatch spacing. As noted by Saghaian et al. [152], this parameter can affect the microstructure, texture, and thermomechanical properties of alloys produced by LBAM (Figure 11). A complex relationship between hatch spacing and dwell time should also be considered when treating this as a variable parameter [2, 153].

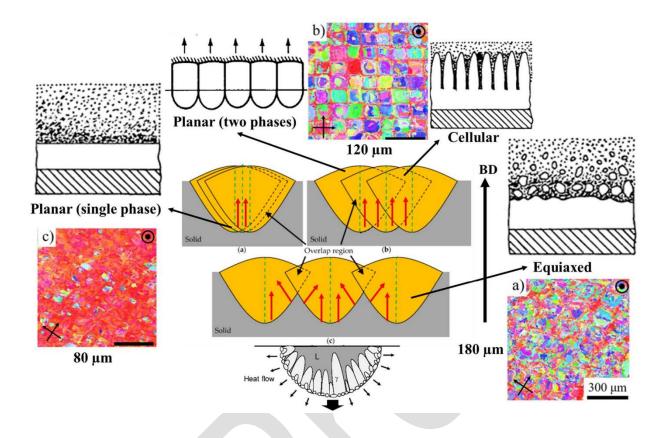


Fig. 11. Effect of hatch spacing on the microstructure in LPBF [152]

#### 5. Defects (Pores and Cracks)

Parts produced via AM generally exhibit varying quality depending on the process parameters and/or feedstock used. The main defects observed in these parts are pores and cracks (Figure 12).

#### 5.1. Cracks

The formation of cracks at the initial stage largely depends on the selection of raw materials. For this reason, weldable materials are predominantly used in AM processes. At later stages, process parameters significantly influence crack formation (Figure 13). The types of cracks reported in the literature for NBSAs are as follows:

#### • Solidification Cracking

When the volume fraction of the solid phase ranges between 0.7 and 0.9, the remaining liquid struggles to flow through the dendritic structure. Shrinkage of dendrites can lead to the formation of pores, which subsequently result in solidification cracks. This issue is particularly severe in non-weldable superalloys due to their pronounced segregation behavior. Also, solidification cracking is the most common type of cracking observed in solution-strengthened superalloys such as IN625 and Hastelloy X. These alloys contain elements like Hf, Nb, Mo, and C, which facilitate the formation of carbides and Laves phases. These phases exhibit low eutectic temperatures, which extend the mushy zone and increase the material's susceptibility to cracking.

## • Liquation Cracking

Liquation cracking occurs when rapid heating prevents secondary phases from dissolving into the matrix, causing them to transition directly into a liquid phase. This liquid phase is unable to withstand the stresses induced by thermal contraction. Liquation cracking typically originates from eutectic phases, such as  $\gamma/\gamma'$ , and is most commonly observed in the heat-affected zone (HAZ). The presence of elements such as Si, Zr, and B can increase the susceptibility to liquation cracking in AM.

### • Strain-Age Cracking (SAC) and Ductility-Dip Cracking (DDC)

These types of cracks occur in the solid state. Strain-age cracking (SAC) typically occurs in  $\gamma$ ' strengthened superalloys with high Al and Ti content, which rely on precipitation strengthening. Ductility-dip cracking (DDC) has a more complex mechanism and is sometimes categorized as a form of liquation cracking or SAC.

To address the sensitivity of raw materials to crack defects, feedstock is typically selected from weldable NBSAs (Section III.2). However, in AM processes, due to rapid liquation and solidification, even weldable materials can become prone to cracking. Some researchers attribute cracks primarily to the energy input into the melt pool. A more comprehensive approach, however, involves examining the individual effects of each process parameter on defect formation. The range of process parameters influences the melt pool shape, liquation and solidification rates, and the heat-affected zone (HAZ). While these factors have been extensively studied, the results of current research are often inconsistent. For example, some studies report that increasing scan speed leads to higher crack density, while others present contradictory findings. Similar discrepancies are observed for laser power. One potential gap in the research is the limited consideration of element diffusion during AM processes. While only a few studies have investigated this aspect, some suggest that the presence of specific elements may significantly contribute to crack formation. Additionally, certain studies have noted the accumulation of elements, such as aluminum, around cracks, though these findings were incidental rather than the primary focus of investigation. Another cause of conflicting results across studies could be the limited parameter ranges considered and the lack of consistency in comparing results. For instance, the impact of increasing scan speed at low energy inputs may differ significantly from the same change at high energy levels. Table 7 presents the typical parameter ranges studied in the LPBF method, the most common AM technique for NBSAs.

#### **5.2. Pores**

It is common to observe pores in AM-produced parts after evaluation (Figure 13a). The origins of these pores can be categorized into two main types. The first type of pores results from LOF. LOF defects occur when the laser source fails to adequately melt the substrate or raw material. The

second type originates from entrapped gases within the sample. These gases can form under various conditions. For instance, improper production methods of AM powder may introduce gases into the raw material, which are subsequently transferred to the part during the AM process. Additionally, some gases are produced due to the evaporation of volatile elements in the melt pool. This type of pore can be minimized by optimizing the process parameters. High energy input to the melt pool, achieved by decreasing the scan speed or increasing laser power, alters the melt pool shape. Specifically, increasing the energy input transforms the melt pool shape from a teardrop to an elliptical form. This change increases the width-to-depth ratio of the melt pool, facilitating the escape of trapped pores.

As mentioned before, some studies associate defect formation with the energy input to the melt pool, while others focus on the individual effects of process parameters. The values of process parameters can vary widely, as shown in Table 7. Most investigations have explored the influence of parameters in the high scan speed range. However, at low scan speeds, even small changes in speed can lead to significant variations in energy input, as indicated by Equation 1. This highlights a research gap in the low scan speed range that warrants further investigation.

$$E = \frac{P}{v \times t \times h} \tag{1}$$

Where P is the laser power (W), v is the scan speed (mm/s), t is the layer thickness (mm), and h is the hatch spacing (mm).

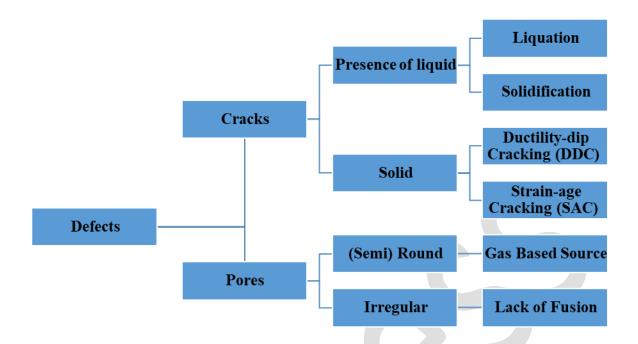
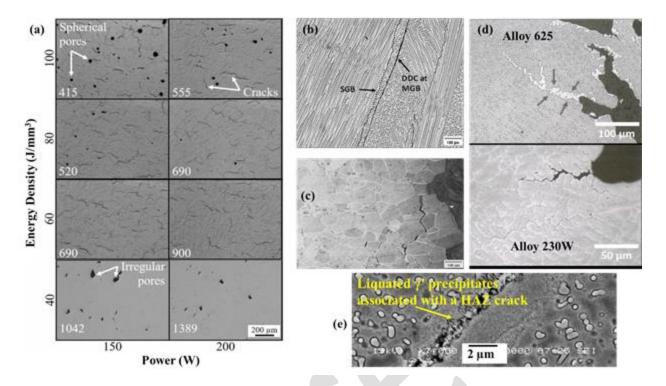


Figure 12. Common defects in LBAM of NBSAs [36, 109, 139]



**Figure 13.** (a) Various pores, including spherical and irregular types, and cracks observed under different process parameters in LPBF of IN625; white numbers indicate scan speeds [154], (b) DDC in Ni-based weld metal along a migrated grain boundary (MGB); the dotted line represents the solidification grain boundary (SGB) [155], (c) SAC: intergranular strain-assisted cracking (SAC) in the simulated HAZ of Waspaloy [155], (d) Solidification cracking; the white area indicated by arrows is a Nb-rich eutectic [156], (e) Liquation cracking in the HAZ of Ni-based superalloy welds [157]

Table 7. Common range of LPBF process parameters

| Scan Speed   | Power      | Layer     | Weldability        | Descriptions        | Reference |
|--------------|------------|-----------|--------------------|---------------------|-----------|
| Range        | Range      | Thickness | (Superalloy)       |                     |           |
|              | C          |           | , 1                |                     |           |
| 725 to 875   | 169 to 195 | 20 μm     | 625                | Hatch= 90 to 110 μm | [189,184] |
| mm/s         | W          |           |                    |                     |           |
| 320 to 1250  | 156 to 195 | 20 to 40  | IN625              | Hatch= 80 to 100    | [١٤٠]     |
| mm/s         | W          | μm        |                    | μm                  |           |
| 1000 to 1300 | 100 to 170 | 20 μm     | non-               | Hatch= 50 to 70 μm  | [191]     |
| mm/s         | w          |           | weldable           |                     |           |
|              |            |           | MAD542 and         |                     |           |
|              |            |           | ME3                |                     |           |
| 1000 to 1900 | 170 to 195 | 20 μm     | Rene 80            | Hatch= 30 to 80 μm  | [197]     |
| mm/s         | w          |           | non-               | ,                   |           |
|              |            |           | weldable           |                     |           |
| 2800 to 3200 | 170 to 220 | 20 μm     | 247 LC non-        | Hatch= 20 to 40 μm  | [154]     |
| mm/s         | W          |           | weldable           |                     |           |
| 200 to 2200  | 125 to 350 | 30 µm     | IN718,             | Hatch= 60 and 120   | [194]     |
| mm/s         | w          |           | weldable           | μm                  |           |
| 800 mm/s     | 195 w      | 20 μm     | IN625,<br>weldable | Hatch= 100 μm,      | [190]     |
|              |            |           | ,, claudic         | changing in Scan    |           |
|              |            |           |                    | strategy            |           |
| 500 to 1000  | 170w       | 30 μm     | IN718              | 30% overlap         | [۶۲]      |
| mm/s         |            |           |                    | between passes,     |           |

|                  |            |          |             | Hatch= 56 µm,             |       |
|------------------|------------|----------|-------------|---------------------------|-------|
|                  |            |          |             | παιείι— 30 μπι,           |       |
|                  |            |          |             | laser spot= 80 µm         |       |
|                  |            |          |             |                           |       |
| 800 to 1400      | 200 w      | 30 µm    | K418,       | Hatch= 70 μm              | [199] |
|                  |            |          | medium      |                           |       |
| mm/s             |            |          | weldability | Al+Ti=4.3%                |       |
| 400.2000         | 100.000    | 20       | G) fo to to |                           | 54 43 |
| 400-2000<br>mm/s | 100-200 w  | 20 µm    | CM247 LC    | Laser spot: 150µm         | [١٠٩] |
| 11111/8          |            |          | non-        |                           |       |
|                  |            |          | weldable,   |                           |       |
|                  |            |          | weidabie,   |                           |       |
|                  |            |          | directional |                           |       |
|                  |            |          |             |                           |       |
|                  |            |          | solidified  |                           |       |
|                  |            |          |             |                           |       |
| 500-1500         | 150-350w   | 30-62 μm | IN718       | (Optimized)<br>P=245-255w | [17]  |
| mm/s             |            |          | recoldable  | V = 850-1000  mm/s        |       |
| mm/s             |            |          | weldable    | Hatch: 100-110 μm         |       |
|                  |            |          |             | Layer thickness: 43-      |       |
|                  |            |          |             | 48 μm                     |       |
|                  |            |          |             | Beam diameter: 100        |       |
|                  |            |          |             | μm                        |       |
|                  |            |          |             | μιιι                      |       |
| 150 to 250       | 400 to 600 | 1.4 mm   | CMSX-4      | Selective laser           | [197] |
|                  |            |          |             | epitaxy (SLE),            |       |
| mm/s             | W          |          | non-        | Layer thickness and       |       |
|                  |            |          | weldable    | hatch: 1 to 2 mm          |       |

## 6. A Commentary on Additive Manufacturing of Gamma-Prime Inducing NBSAs

As mentioned earlier, the superalloys with a sum of aluminum and titanium greater than 6.4% are sorted as non-weldable alloys. Two conventional grades of this non-weldable group are IN738 and CMSX-4. Defectless joining in these alloys without using preheat is so difficult practically. In these alloys, strain-age cracks are created in the specimen due to the formation of the gamma-prime phase. In order to solve the problem, preheat facilities have been designed for some new additive manufacturing machines. For the proper joining of superalloys like IN738, the preheat

temperature is usually set between 650°C to 800°C; applying this temperature makes the process hard in structure. Despite complex manufacturing, some advantages include small gamma primes and more homogenous distribution in the AM-developed parts.

Conversely, the manufacturability of components with intricate internal cooling channels, such as new turbine blades, has catalyzed research efforts toward the exploration of additive manufacturing techniques for these particular alloys. Moreover, in single crystal grades, including CMSX-4, AM-developed parts are prone to form columnar structure, and it is possible to customize the structure to a large extent. Therefore, much research has been carried out regarding the effect of the process parameters on the type of solidifying structure and columnarization of the grains [111, 168-179].

#### 7. Microscopy

Two important items in the non-weldable superalloys are the size and distribution of the gamma prime particles. As mentioned, the present gamma primes in the AM-developed parts are smaller than those in the cast parts. In order to receive an acceptable image of the existing gamma primes in the parts, it is necessary to use a proper etchant solution to prepare them.

• Marble's solution is proper because it preferentially attacks the gamma prime phase and leaves the residual gamma-matrix unaffected. The composition of Marble's solution is as follows [180]:

50 ml HCl, 50 ml H<sub>2</sub>O, and 10 mg CuSO<sub>4</sub>

• Another conventional reagent has the following composition [181]:

10 ml HNO<sub>3</sub>, 50 ml HCl, and 60 ml glycerol

• For a proper electrical etch, the following composition may be used. It is noteworthy that the process is carried out in 6 V for about 5 s [182-186].

 $12 \text{ ml } H_3PO_4 + 40 \text{ ml } HNO_3 + 48 \text{ ml } H_2SO_4$ 

### 8. Applicable Mechanical Tests

In general, numerous mechanical tests may be used to evaluate the mechanical properties of one specimen made of NBSAs. Different types of tests are employed to determine the working conditions of the sample. For instance, NBSAs often has elevated/high-temperature applications; therefore, the creep test is highly important. Fatigue [187], room and high-temperature tensile strength [147, 188-194], punch and tensile creep [195-203], relaxation, hardness, and microhardness [147, 204, 205] tests may be specimen evaluation metrics. Some tests are used to predict the material properties in another test. For example, if the creep test is not practically possible, the relaxation test can be employed to have a total assumption of the material properties [115].

#### 9. Improvement of AM-Developed Part Properties by HIP Treatment

If the LPBF- and DLD-developed parts are subjected to heat treatment or HIP, the final properties will be comparable to the parts produced via conventional methods. Primarily for AM-developed parts, the same conventional heat treatment is applied to casting parts. However, it is evident that the traditional heat treatment process is an objective initiation point and needs to be improved and adapted for AM-developed parts [35, 206]. Therefore, much research is needed in this regard.

By employing HIP, nearly all cracks are removed from susceptible alloys, such as IN738, and the porosities become small. By this method, only the internal cracks (the cracks connected to the surface remain) are eliminated. Machining can be employed to eliminate near-surface cracks with

interconnections; however, this approach may be infeasible in instances where the sample's geometry precludes such intervention [109, 141].

#### 10. Advantages of Segregation Reduction and Homogeneity Enhancement in AM

Another advantage of AM-developed parts is their improved chemical homogeneity, which is attributed to the small melt pool size in AM processes. Due to rapid solidification and limited melt volume, alloying elements have insufficient time to segregate during solidification. In contrast, cast specimens typically exhibit greater elemental segregation. In casting, alloying elements are pushed toward the remaining melt as solidification progresses, resulting in significant composition gradients between the initial and final solidified regions [34, 141, 207]. Rickenbacher et al. [141] illustrated this behavior by plotting the composition variation range in cast specimens and demonstrated that it is significantly higher than in AM-developed counterparts.

#### 11. New Materials

Nowadays, due to the previously mentioned challenges in processing  $\gamma$ '-inducing NBSAs, studies have been conducted on modified materials. NBSAs were conventionally produced by casting; therefore, their chemical compositions were tailored to achieve the required properties after casting. In recent years, the same cast alloys have been used in AM processes. Some materials, such as IN738, retain their strength and performance at high temperatures and in extreme environments due to the presence of the  $\gamma$ ' phase [208]. However, as discussed earlier in Section III.2, their inherent properties lead to poor weldability and, consequently, poor AM processability. Recent studies have shown that crack-free LBAM of these traditionally non-weldable alloys is possible, provided that process parameters are carefully selected [209]. These parameters often require re-optimization for each new experimental condition, such as variations in geometry,

powder composition, or particle size [210]. However, in industrial applications with complex geometries, achieving crack-free production remains a significant challenge. To address these issues, efforts have been made to reduce the content of  $\gamma$ '-forming elements such as Al and Ti. In addition, some studies have aimed to modify the chemical composition to narrow the solidification range, thereby minimizing the risk of solidification cracking. ABD-900AM is one such newly designed superalloy. After proper heat treatment, it forms approximately 35%  $\gamma$ ' by volume. In these alloys, the reduced  $\gamma$ ' strengthening is compensated by the addition of solid-solution strengthening elements such as W and Mo, which strengthen the  $\gamma$  phase. The resulting mechanical properties at high temperatures are comparable to those of IN738 and IN739, which are commonly used in turbine blades and hot-section nozzles of gas turbines [26, 211-216].

#### **IV. Conclusion**

The review begins by addressing the progress and development of LBAM (DLD and LPBF), providing a foundation for understanding the evolving capabilities and limitations of these methods in industry. By exploring the development trajectory of LBAM, we gain deeper insight into how LPBF and DLD can be optimized and applied across various applications. The second half of the review focuses on a case study of nickel alloys, emphasizing how the knowledge of LBAM's evolution helps to better understand the potential and challenges of these techniques in the context of specific materials.

The findings of this review indicate that, LPBF and DLD are transforming the manufacturing landscape for NBSAs, offering significant advantages for the fabrication and repair of complex components. Traditional methods like casting and wrought processing are constrained by geometric limitations, long lead times, and difficulties in defect control. In contrast, LPBF and DLD can produce net-shape parts with intricate geometries, reducing material waste and offering localized heat input for improved microstructures. This review has highlighted key aspects of the processes, including powder characteristics, laser-material interaction, thermal cycling, and microstructural evolution, with a particular focus on how processing parameters influence final part quality.

The processability of various NBSAs in AM has been a central theme, with weldable alloys like IN625 being more amenable to LPBF and DLD, while more challenging alloys such as IN718 and non-weldable NBSAs like IN738 and CMSX-4 require tailored strategies to overcome processing difficulties. Moreover, the review underscores the need for developing AM-specific post-processing treatments, as traditional methods do not effectively address the unique microstructures of AM-produced components.

Emerging trends such as in-situ monitoring, machine learning-based optimization, and the development of crack-resistant alloys are essential for advancing AM's reliability and repeatability. As research progresses, AM will continue to enable the production and repair of advanced NBSA components with customized performance, offering promising implications for sectors such as aerospace and power generation. The continued exploration of processing strategies, alloy design, and microstructure-property relationships will be crucial to fully realizing the potential of LPBF and DLD in industrial applications.

#### **Conflict of Interest**

All co-authors have seen and agree with the contents of the manuscript and there is no financial interest to report.

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# Glossary

AM: Additive Manufacturing

LBAM: Laser-Based Additive Manufacturing

LPBF: Laser Powder Bed Fusion

NBSAs: Nickel-Based Superalloys

HAZ: Heat-Affected Zone

SLM: Selective Laser Melting

DLD: Direct Laser Deposition

**DED: Directed Energy Deposition** 

EBPBF: Electron Beam Powder Bed Fusion

HIP: Hot Isostatic Pressing